

Investigation of Transient Regime Transition in Inductive Plasma Generators

Robin Georg¹, Ashley Chadwick², Bassam Dally³, Georg Herdrich⁴, Francesco Romano⁴, Mirko Magarotto⁵, and Daniele Pavarin⁵

¹School of Mechanical Engineering, University of Adelaide, Adelaide, Australia
³Centre for Energy Technology, University of Adelaide, Adelaide, Australia
⁵Department of Industrial Engineering (DII), University of Padova, Padova, Italy

²German Aerospace Center (DLR), Stuttgart, Germany
⁴Institute for Space Systems (IRS), University of Stuttgart, Stuttgart, Germany
E-mail: robin.georg@adelaide.edu.au



Motivation

Electrodeless EP using IPG

- Improves propellant flexibility and lifetime
- Enables new operating concepts:
 - In-situ resource utilisation (ISRU)
 - Atmosphere-breathing electric propulsion (ABEP)

Objective

Develop an experimental technique that:

- Investigates regime transition, to control propulsion characteristics
- Is non-intrusive, to avoid disturbing conditions
- Is time-resolved, to capture effects due to non-steady power supply or modulation

Theory

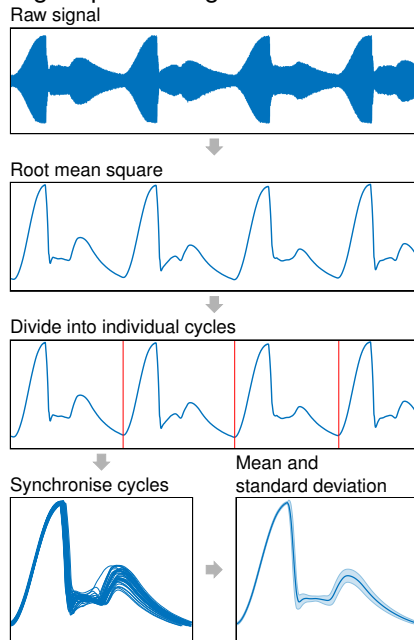
- Antenna current → antenna-plasma coupling
- Magnetic field strength → function of antenna current and plasma currents
- Light emission in visible range → skin depth, location of plasma current

Experiment

3 channels captured simultaneously at 5 MHz:

- HOKA probe → antenna current
- Linear stage → radially-resolved:
 - Magnetometer → axial magnetic field strength
 - Photodiode → light emission in visible range

Signal processing:



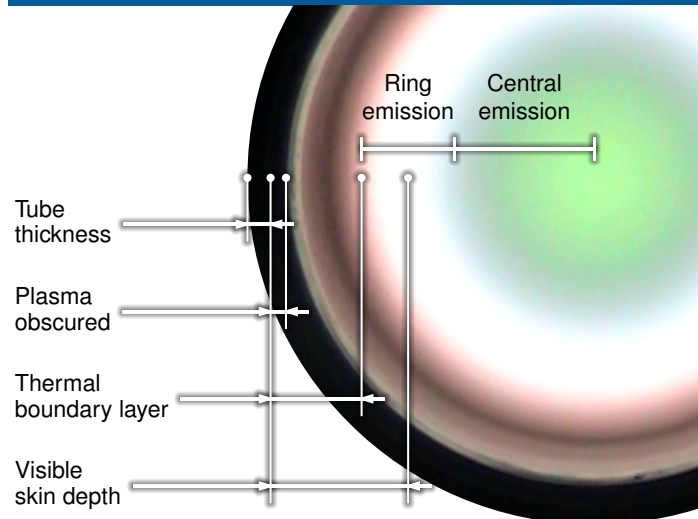
Conditions:

- $f_{\text{driven}} \approx 586 \text{ kHz}$
- $f_{\text{cycle}} \approx 300 \text{ Hz}$
- $p_{\text{tank}} = 28.5 \text{ Pa}$
- $p_{\text{injector}} = 5.8 \text{ kPa}$
- $\dot{m}_{\text{gas}} = 2.7 \text{ g/s}$
- Capacitive and inductive regimes

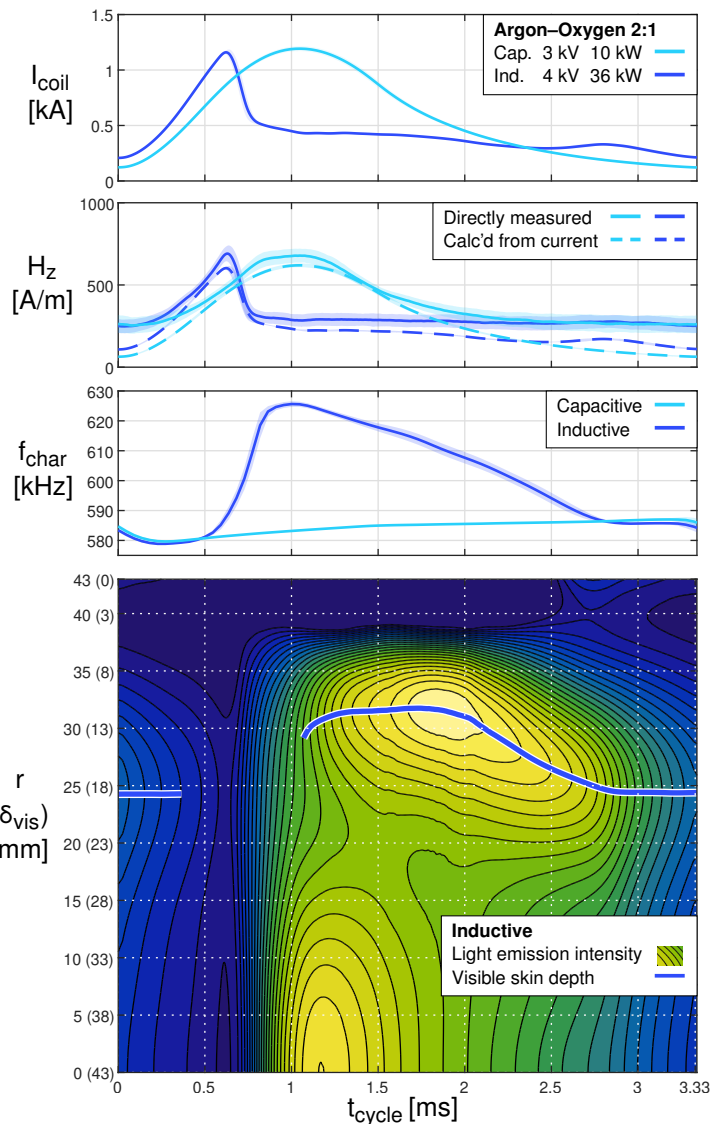
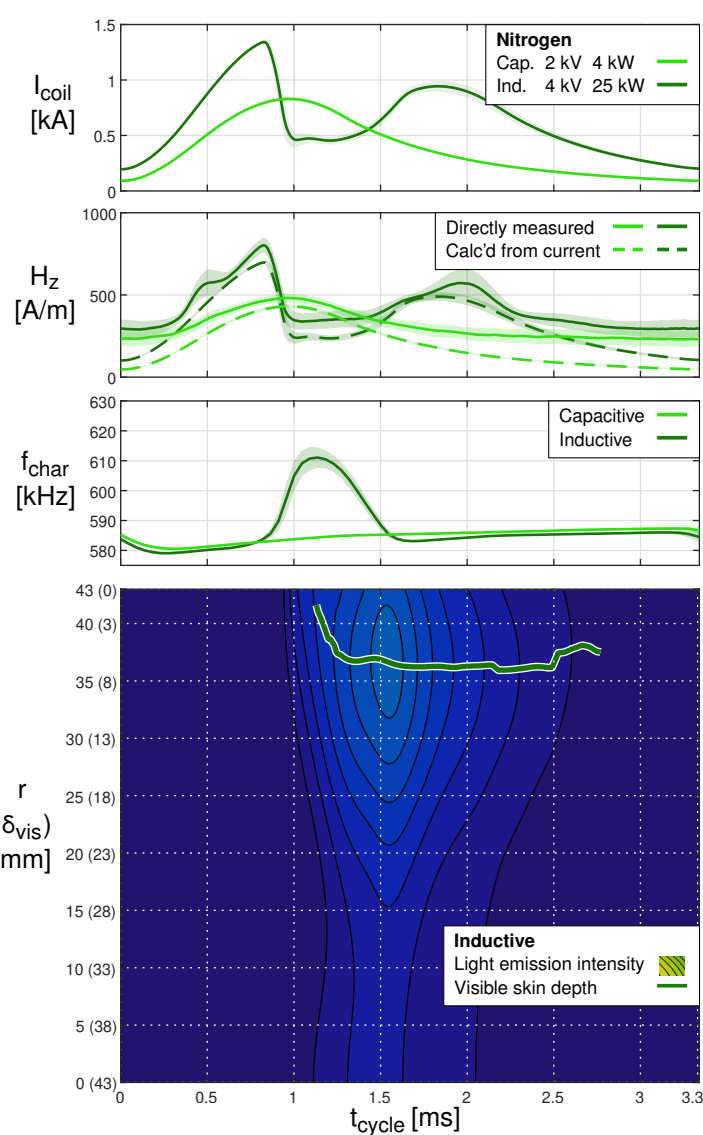
Gases tested:

- Nitrogen
- Argon-oxygen (2:1 by volume)

Light Emission Distribution



Preliminary Results



- Antenna current and characteristic frequency
 - Clearly identify regime transition due to development of plasma current
 - Duration of inductive coupling affected by propellant choice
- Measured and calculated axial magnetic field agree
 - Plasma current has little effect, likely located downstream far from probe
- Argon-oxygen: ring and central emission appear at different times

Conclusions

- Identification of regime transition, and plasma current behaviour, possible at time-scale of discharge cycle
- Plasma current appears to be localised in small, downstream region
- Numerical models may need to consider transient behaviour

IPG7 and Experimental Setup

